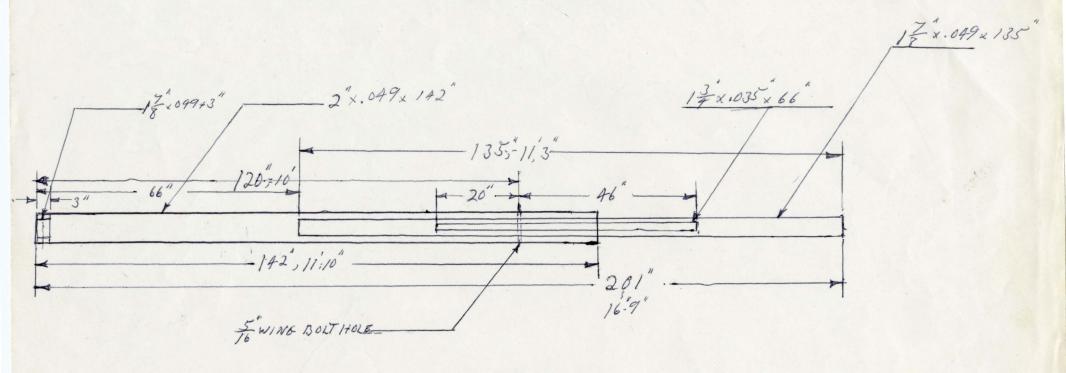
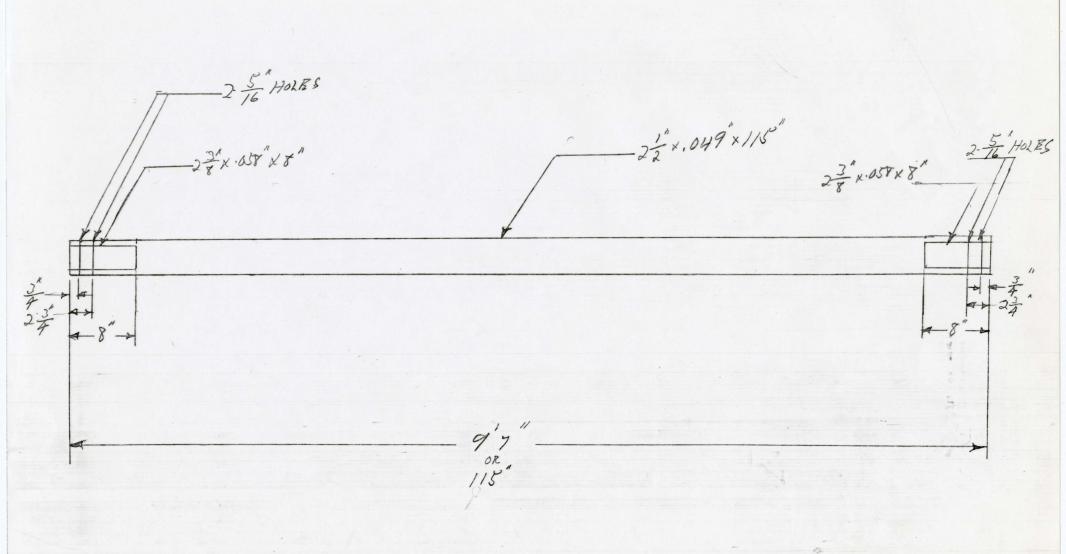
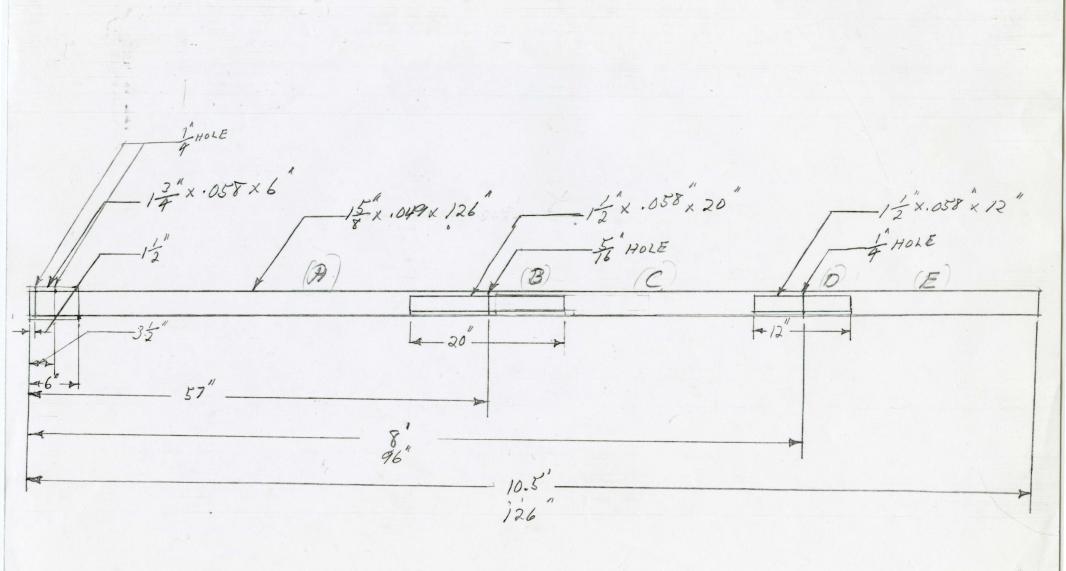
SUGGRET TO REPLACE WITH PANG GLIDER SENSOR SID LEADING EDGE EQUIVARNT GRAPHITE SPAR TAPEARD FROM 2500 TO 13" (2) REQUIRED SCALR = 1' WT= 9.2165/2E, OR 18.4165 FOR(2)



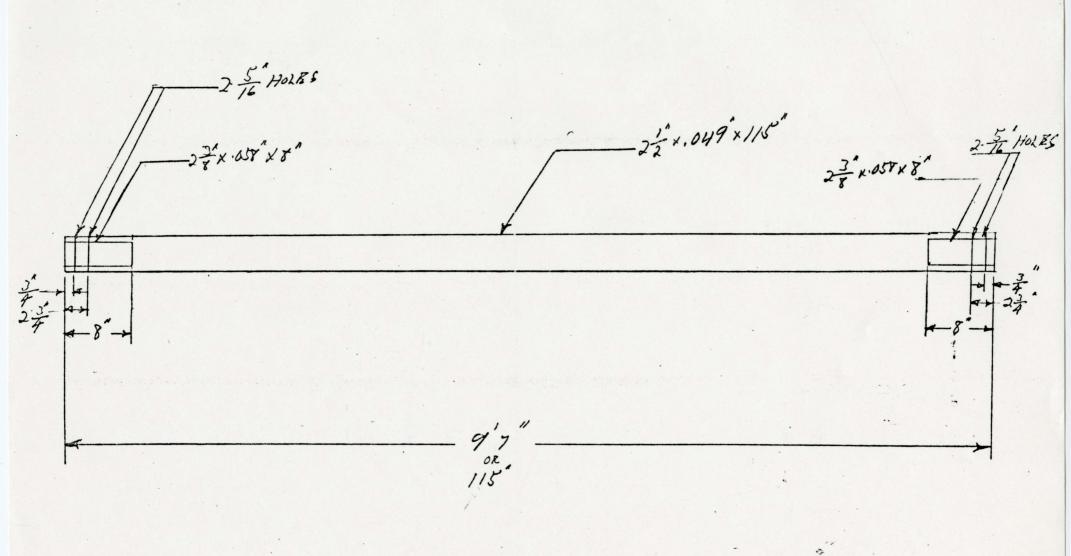
SENSOR-510-CROSSTUBE 2- REQUIRED



SENSOR 510 KEEL 1-REQUIRED



SENSOR-510-CROSSTUUR 2- REQUIRED



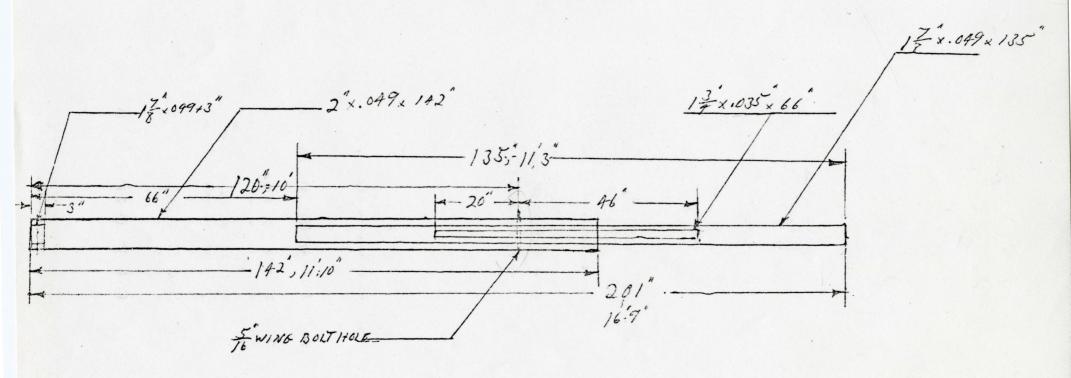
SUCCEST TO REPLOCE WITH PANG GLIDER SENSOR SID LEADING EDGE

EQUIVARNT GRAPHITE SPAR

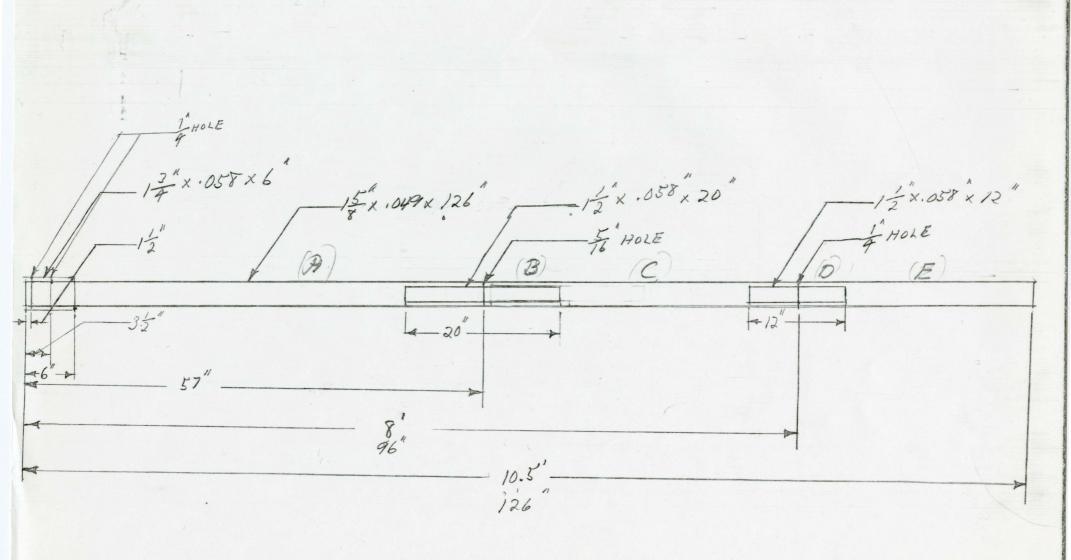
TAPEARD FROM 2500 TO 13"

(2) REQUIRED

SCALFE =1' WT = 9.2165/2E, OR 17.4165 FOR(2)



SENSOR 510 KEEL 1-REQUIRED



I have been working for the past two years with a graduate student, Craig Douglas, on graphite designed tubing for my Sensor 210 D-165, with Bob Trampenau's permission of course. We had about \$3000 worth of graphite tubing. It was 2" and 1 1/2" X .050" wall thickness, 6' lengths, and spun in different weaves from 5°-90°, which governs the stiffness or flexibility of the tubes. With Bob's recomendation we used the 2" on the leading edges and 1 1/2" on the keel and cross tube. We did bending tests on all the different angles the tubes were spun at. (Results are enclosed with this material). We found that the 25° graphite was about the same stiffness as the Aluminum. We used 25° on the forward keel, 5° on aft keel, 25° on the crossmember and 30° on the 2" leading edges. The tubes were sleeved and bushed wherever there was a bolt or splice with stainless steel bushing or anodized aluminum sleeves. The tubes were joined by sleeves with 2 ton 1/2 hour epoxy glue.

The College of Engineering has allowed me to work on this project for the past year, during working hours. Since they paid for all the tubing, I furnished the money for all the new hardware (sail and all anodized sleeves and other parts cost me \$550) First, I replaced the keel with graphite on my glider and flew it, next the crossmember, and finally the leading edges. After all the tubes were replaced I bought a new white sail, fitting it with all new parts. I then put all the aluminum back onto the old glider. It was a step by step operation. The weight reduction was not as much as I expected because the leading edges were 1 5/8" and about 1 1/2" replaced by 2" graphite. Tube for tube it is just about 40% lighter and twice as strong. In tension tests the bolts pulled right through the Aluminum sleeves that were epoxied onto the graphite. The glider at the start weighed 51 pounds, while after conversion it weighed only 45 pounds. The sail weight was 10 pounds, the control bar, king post, deflexor, define tips and battens about 10 pounds. I have ordered a control bar, king post and defined tips in graphite also, and they should reduce the weight by about 5 pounds more. The College of Engineering is also paying for more graphite tubing, 1 7/8" and 1 5/8" for a deflexorless, Sensor 510.

TUBING COMPARISON

Keel-aluminum Keel-graphite

Weight --- 1983 grams or 4.37 pounds Weight---1180 grams or 2.60 pounds

Cross tube-aluminum Cross tube-graphite

Weight--- 1297 grams or 2.86 pounds Weight--- 877 grams or 1.93 pounds

Leading edge

Graphite (2")

Aluminum (1 5/8"& 1 1/2") Weight--- 2782 grams or 6.13 pounds Weight--- 2603 grams or 5.73 pounds

GRAPHITE TESTS

1 1 2" X .050"-.006 ±5°

'eight	Deflection	E= PL.
41bs-6 oz.	1.560	48 y Max π Ravg. t
124 lbs-0 oz.	1.604 1.660	
4 lbs	1.717	77 71 07 77 706
84 1bs 104 1bs	1.776 1.810	E= 14.31 X 10°
124 lbs-6 oz.	1.858	

1 1/2" X .050"-.004 ±45°

reight	Deflection				•		
0	1.5)1						
4 lbs-6 oz.	1.626				3		
24 lbs	1.862		E=	-	PL		
44 lbs .	2.128			4.8	y Maxx	R3 avg	t
64 1bs	2.370				6		
84 lbs	2.653		E=	2.7	X ·10		
104. lbs	2.964		•				

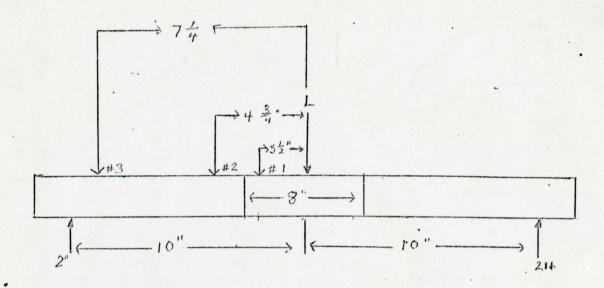
1 1/2" X .050"-.003 ± 25°

Weight	Deflection	
0 4 lbs 6 oz.	1.540 1.570	E= PL3
24 lbs 6 oz.	1.630 1.724	48 y Max 7 Ravg t
64 lbs 6 oz. 84 lbs 6 oz. 104 lbs 6 oz.	1.800 1.876 1.975	E= 8.52 X 10

ALUMINUM TESTS

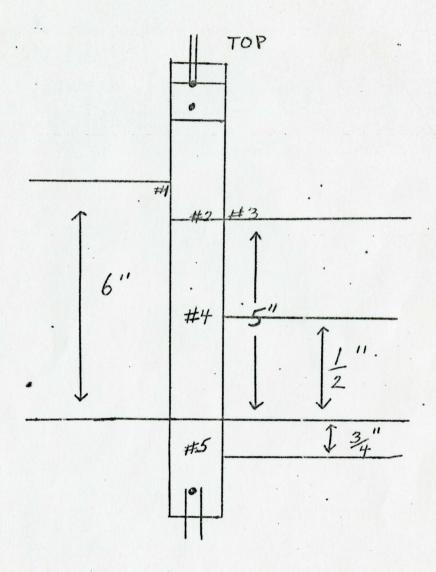
1 1/2" X .040"

Weight	Deflection			
0	1.573			
4 lbs 6 oz.	1.610	77	3	
24 lbs 6 oz.	1.690	E=	PL	Market Market
44 lbs	1.768	48	y Max n R avg	t
64 lbs 84 lbs	1.834 1.927			
104 lbs	1.987	T= 8	95 X 10	
204 200	1.0 10 1	기계 시간 아이를 하게 된 것이다. 프린	75 A 10	



Gage No.	Reading(lbs)	Strain or comp.	deflection
	50	135	.0155
	100	268	.038
"	150	405	.0615
# 1	200	223	.0875
	250	150	.13
	300	90	.43
	500	94	.76
	50	532	
	100	1080	
	150	1652	
# 2	200	2020	
	250	2549	
	300	3104	
	500	6100	
	50	320	
	100	665	
	150	1010	
<i>#</i> 3	200	1360	
#)	250	1693	
	300	2028	
	500	3430	

Gage No.	. Reading (lbs)	Strain	Micro inches/inch
	50		21.01.00) 11.011
		35	
# 1	100	54 78	
π 1	150		
	200	100	1500 678
	250	123	1500 070
	50 .	14 .	
	100	40	
# 2	150	67	
	200	80 .	
	250	. 88	1500 512
	50	30	
	100	52	
# 3	150	78	
	200	95	
	250	118	1500 660
	50	22	
	100	24	
# 4	150	25	
	200	24	
	250	22	1500 306
	50	10	
	100	10	
# 5	150	10	
	200	10	
	250	10	1500 178



January 27, 1980

I test flew a glider with a new 25 front keel and 5 aft keel, with 25 cross tube. I had two good long flights and the glider did feel 4 lbs lighter.

February 9, 1980

I test flew a glider with new graphite leading edges. I had a good flight and it held up good.

SIZNSOR SID-HEEL- I REQUIRED EL = 34/64 D4-14 EAL = 10 x 10 6 PSI) (D4-d4) = 10 (D4 d4) 34(D4-64)=+ 10 (1.625) 4 (1.527) 4 = K 10 (6.973) - (5.437) = 11.53.6(34) = 14=.452 2.4 = D4.-K X = .452 d4 = (1.625) 4-.452 d7 = 6.973 -452 = 6.521. d = (6.521) 4= 1.598 OD. GR. - 1.625 = 1.598 d = 1.598 [WALL TH = .0135" FOR D-C

FOR SEC BAD.

10-(D4-2")=K

10 (1.625) 4- (1.411) 4=x

10(6.973)-(3.964)=+

1º (3.009) = A

1.885 = 4

d" = 54-A

d 4= (1.625) 4-, 885

d+ = 6.973 - 1845 = 6.118

d = (6.118)4 =

d = 1.573"

00.6n = 1.625. 1.0. -1.573

WALL TK = . 026 "/ FOR SACBED

$$\frac{SENSORSON-HEEZ-1 REQUIRED}{EII = 344 (64)} D^4-d^4) \qquad (E_{GR} = 344 / 16^6 - PSI)$$

$$\frac{D^4-d^4}{64} = \frac{10}{34} (D^4-d^4) \qquad (E_{GR} = 10 \times 18^6 PSI)$$

$$\frac{10}{34} D^4-d^4) = K$$

$$\frac{10}{34} (1.629) (1.527) = K$$

$$\frac{10}{34} (6.973) - (5.437) = 11.53 \cdot 6 \frac{10}{34} = 14 = .452$$

$$\frac{10}{34} (1.625) - (5.437) = 11.53 \cdot 6 \frac{10}{34} = 14 = .452$$

$$\frac{10}{34} (1.625) - (5.437) = 11.53 \cdot 6 \frac{10}{34} = 14 = .452$$

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$$\frac{10}{34} (1.625) - (5.437) = 11.53 \cdot 6 \frac{10}{34} = 14 = .452$$

$$\frac{10}{34} (1.625) - (4.527) - (4.527) = 11.538$$

$$\frac{10}{34} (1.$$

GRAPHITE TEST WT. AL-VEST DEF. at.

1 Side tube

Braid 4-8 layers \ 40 ends 12K PAN Graphile
40 ends 52CG 150 1/3 x 2

75% graphete 25% gloss

Wught of take 52 300 201" LOA " 1" 2WD, = 4.875 45 (ext. 4.7)

Resin ~ 50 % by weight 250°F Film

adhesive

- wedge on mandrel supped out an outland 3' coasony field in cloth - cut off & will be required

Diameter 1D 1.875

POR BRUDING OF TRUSION MIDDLAC

material Turnished by B 1,90" x'20' ALPIPE -- 63.60 & X & STERL PVC PIPE = 2.9 2 - 12' LRN. 76 56 STOCK =temper the Electrical KER)

CROSS-Side tube 0058 Built to replace 9'7" x 21/200 x,049 altusing 5 layers \$ 40 end 12k PAN graphite \$6=1.9 80 ends Sglus 52GG 150 1/3×2 86=2.5 Weighouf 18 tule E=12.4 406 5 \$ 6 0g W/ renform layer or 5.325 the Weight Sang 3.14# - 37% Wo muset 4 layer of Braid yim the some EI as the above alumnim tube therefore the 5 layers 110/2 guie 25% higher EI. 沙公 Diameter of Take = 21/2" ±020" Ends of take ase reinforced w/ 2 extra layer estimate wall trickens to be .060" Estimate for LE. tuke. weight each 4,7 # 2,1"00 replace 9,2 \$ Bushed with arrow shaft 17 DIA WT = 3. 18/15

white Blaster.

SENSOR SID-HEEL- / REQUIRED EL = 34 64 D4-d" EA = 10 x /0 6 PSI) (D4-d4) = 10 (D4 d4) 34(D4-64)=t 10 (1.625) 4 (1.527) 4 = K 10 (6.973) - (5.437) = 1.53.6(34) = [H=.452] d.4 = D4-K X = . 452 d4 = (1.625)4-.452 d + = 6.973 -452 = 6.521. d = (6.521) 4= 1.598 d = 1.598 10 = 1.598 | WALL TH = .0135" FOR D-C8

04%

FOR SEC BAD.

 $\frac{10}{34}(D^{4}-D^{4})=16$ $\frac{10}{34}(1.625)^{4}-(1.411)^{4}=16$ $\frac{10}{34}(6.973)-(3.964)=16$ $\frac{10}{34}(3.009)=16$ $\frac{10}{34}(3.009)=16$ $\frac{10}{34}(3.009)=16$

a - R - M

d 4= (1.625) - , 885

d+ = 6.973 - 1885 = 6.118

d = (6.118)4 =

d = 1.573"

0D.6n = 1.625. 1.D. -1.573 .052

WALL TK = . 026 "/ FOR SACBED

FOR SEC BAD.

10-(D4-2")=k 10-(1625)4-(1411)4=x

10(6.973)-(3.964)=+

1º (3.009) = A

1.885 = 4

d= 54-+

d 4= (1.625) - , 885

dt = 6.973 - 1885 = 6.118

d = (6.118)4 =

d = 1.573"

006n = 1.625. 1.0. -1.573

WALL TK = .026"/FOR SACBED

FOR SEG BAD.

10 (D4-2")=k 10 (D4-2")=k 10 (1.625)4-(1.411)4=k

10(6.973)-(3.964)=+

134(3.009) = H

1.885 = H

d= = 54-4

d 4= (1.625) +- , 885

d4 = 6.973 - 1885 = 6.118

d = (6.118)4 =

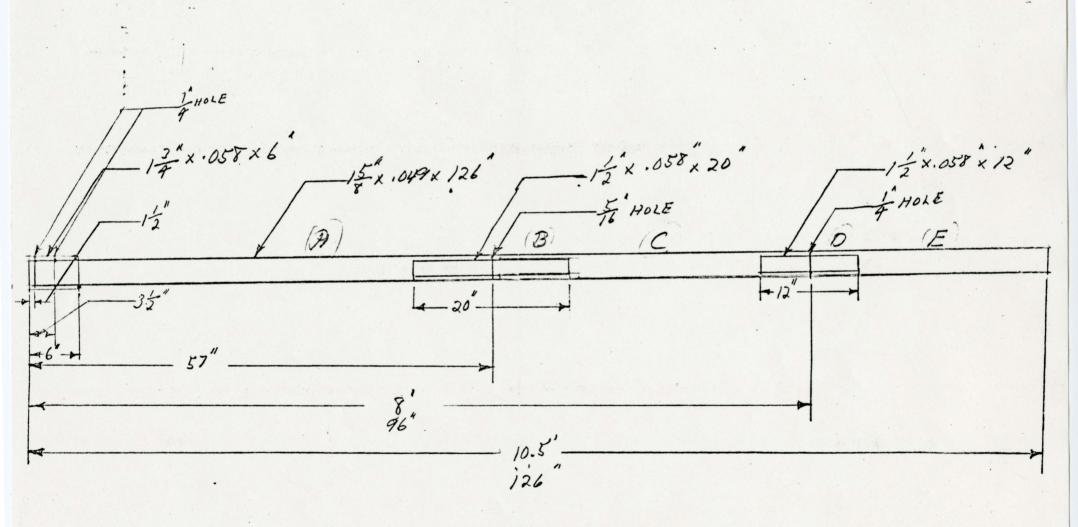
d = 1.573"

00.6n = 1.625. 1.0. -1.573

WALL TK = . 026 "/ FOR SACBED

Cisa Popari lerry trutol Majarie Toberchi's And Finholy
Evic 1s anite Dechare Clouden Brooquel Evic Wisnochi Silven Healy V

SENSOR SID KEEL

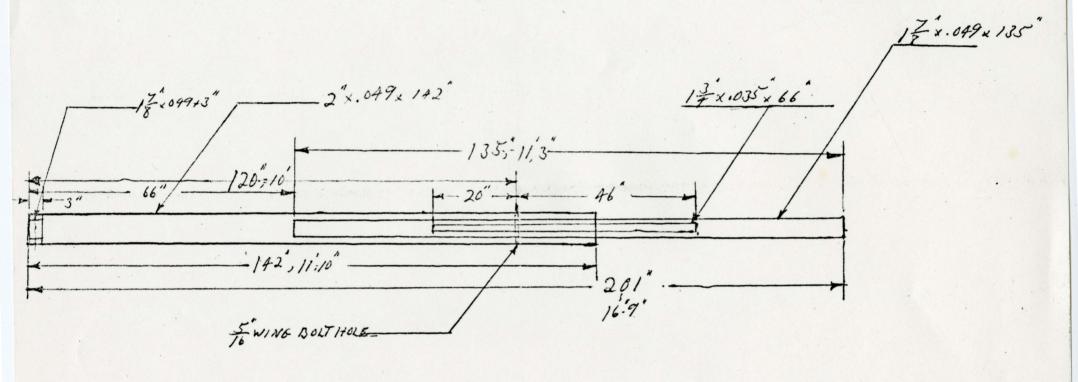


TAPERED FROM 2500 TO 13"

SUGGEST TO REPIBLE WITH PANG GLIDER SINSOR SID LEADING EDGE

(2) REQUIRED

SUPLE 5 = 1'
WT = 9.2165/LE,
OR 18.4165 FOR(2)



	TENGUL	TRST ON.	1500 x 049 = 450
		STRAIN	
	165	DO MIT.	th 10P
for the	50,	35	Till a land that
	150	78	
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7 1	30	14	6-11 Hz #3
在3——	150	40	5" Tu
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	200	45	1500 (16
		110	1500 660
(An	(50	25	
# gr	,00	24	
	150	25	
	13	24	
	200	22	
			1500 306
	+	10	
	100	10	
	The state of the s	10	
	150	10	
	2 8 9		
<u> </u>		10	
			1500 158

Top fleve let go 1715 165 Botton alm. Folk hole let gor 2/65

768,069 42 O R PL3 48EI Det ymax = 402 I = 17 Rang t $E = \frac{PL^3}{1.355} = \frac{8.52 \times 106}{1.355}$ E = PL3 E = 48 gmax TT Ray t 1.2897 1,8101 8.95 X/06 En= 10x10° p51 < $\frac{2^{1}}{43^{5}} = \frac{2^{1}}{2^{1}} + \frac{10^{4}}{10^{4}} = \frac{10^{4$ 1.858 1951 10/375 1,3 T = 6.49 × 10-2 (2-14.5)

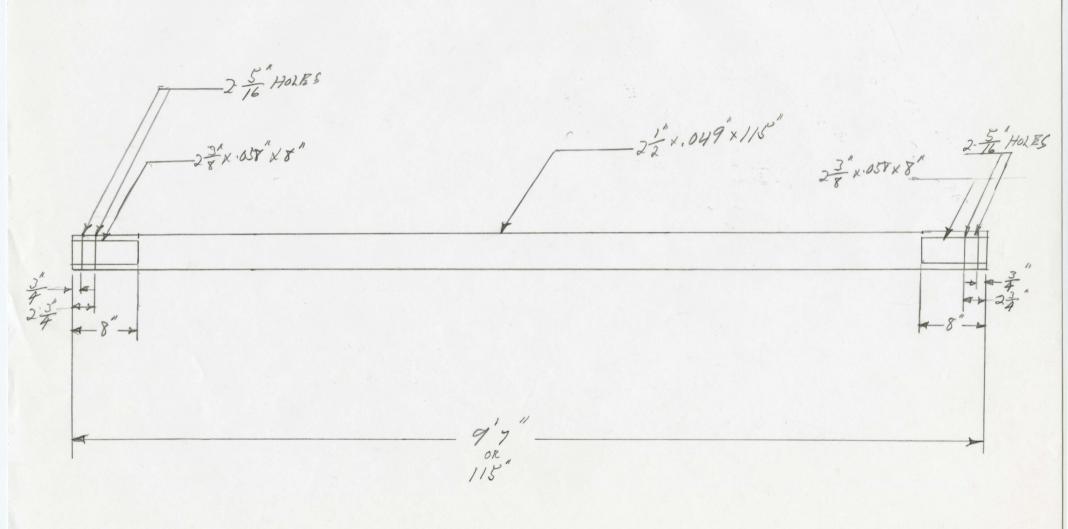
768.069 329.0291

BROKE 1015 1/25 Graphete Bubo broke at 8" sloose 1150 AL

	Bendengtest on 200 4049 - 450	, .
GAL No	2 1 2 1 2 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2	
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42	50 100 1080 1652 2020 2549 300 3104 500 6000	
4.3	320 100 150 1010 1360 1693 300 2028 500 3930	

FORD = 23 SENSON-510 CROSS- YUBE: "2- REQUIRED $\frac{d^{4} \cdot (\lambda' - d^{4})}{(\Delta' - d^{4})} = \frac{10}{34} (D^{4} - d^{4})_{AL}$ $\frac{(\Delta' - d^{4})^{2}}{(\Delta R)^{2}} = \frac{10}{34} (D^{4} - d^{4})_{AL}$ $\frac{10}{34} (D^{4} - d^{4})_{AL}$ 166 = 34 (TT (D4-d 4) 2 - 1 4) = x (25)4 - (2.402)4 =x 39.06-33.62 105.39 71-11.585-11 124 = D4-K K= 1.575 24 = (2.5)4 - 1.584 14= 39.06=1.584=37.476 d = (37.476)4 d = 2.474 x1 8D.GR = 2.500" OD I.D. 2.474 LD THUNDLL = .013"/

SENSOR-510-CROSSTUBLE 2- REQUIRED



SIZNSOR SID-HEEL- / REQUIRED EL = 34/64 D4-d EAL = 10 × 10 6 PSI, (D4-d4) = 10 (D4-d4) 34(D4-64)=x 10 (1.625)4 (1.527)4 = K 10 (6.973) - (5.437) = 11.53.6(34) = [H=.452] d4 = (1.625)4-.452 d + = 6.973 -452 = 6.521. d = (6.521) 4= 1.598 OD. GA. - 1.625 d = 1.598

WALL THE :0135" FOR D-CE SECTION

046

SENSOR SID-HEEL- I REQUIRED E = 34x 10 - PSI. Et = 34/64 D4-d4) EAR = 10×10 6 PSI) (D4-d4) = 10 (D4 d4) 34(04-64)=+ 10 (1.625) 4 (1.527) 4 = K 10-(6.973)-(5.437)=4.53.6(34)= [H=.452] 1.4 = D4.-K X = . 452 d4 = (1.625)+-.452 d = 6.973 -452 = 6.521. d = (6.521) 4= 1.598 OD. Gn. = 1.625" d = 10598 10 = 1.598 .027 WALL TH = .0135" FOR A-CEE SECTION

2.1=3.75 16.4 12 - 5.75 16. GRAPATE. 75-80/ 8-6205 14 200.00/ (Beller impact) 6 20° CRNSOR CIO Botterns - SHAPE Leading Edge \$ 189,00 Conflete. BOTTOM CAN THEY BE PLAT Holo., Silever TIP DOBATER. I set of Batters = x,05 7 7 2 WT. 510 SRUSOR ZEADING EDGE AL = 9.7 185. TIP DOP. 700 V. 035" BATTERN ZODX 1035 + AROW SHAFT Dalleran WT 5.8 1 65 SAUGOR CO Down twee rut 1,5 lbs

SENSOR 10m's 48" -DOUGLE TANG. TRUE APEX-SENSOR 10.5 m's ROOT J-BOLT, & hole. MEASURED SECTION ON. ON L.E. SENSOR 210 = 10'4 NOTE: THERE ARE APEX DPAW-ING FOR . whom colling (.E. Tult. Whit SAIL to ACCOUNT CORRECT for exitionic sleeves SPAR COULT MAINI WINES SPAR 1-5 XDSQ" WITH 900 R Rola 18 out Bons MICPO BALLOOM FILLER MIXED NOSE PLATE. RINGT NYLON PLUB WITH EDOY TOR POLYCSTEIR RESIN-INPLACE. DEBURR FR. AND RADIUS MIX to A MA1936-20A non flowings + MANIA THINTAN'S MIXHURE. SENISOR ZIO = MOSE bott & LLI (3 x.035") WING bOLT 310 = 1000 TIP D-boct 3105 = 1084 AX15 ----FLAZE IN PLUG-IN CABLE SLACKLE OUT BOARD WING SPAZ/="X049" AND SADOLE. CLEVIS PIN 4RING DRILL HOLES, NO-SLOP T ENO-VIEW of complete winds SPAR, Showinds winto boot and AUGIN PELATIONISKIP. CROSS BAR, WINIG SPAR AND CONTRA BAR contion box During toles 1" x083 BASE BAR 1"XOB3 (OP) RIGHT HIGHT S. THANDENHOTTITE

I have been working for the past two years with a graduate student, Craig Douglas, on graphite designed tubing for my Sensor 210 D-165, with Bob Trampenau's permission of course. We had about \$ 3000 worth of graphite tubing. It was 2" and 1 1/2" X .050" wall thickness, 6' lengths, and spun in different weaves from 5° -90°, which governs the stiffness or flexibility of the tubes. With Bob's reccomendation we used the 2" on the leading edges and 1 1/2" on the keel and cross tube. We did bending tests on all the different angles the tubes were spun at. (Results are enclosed with this material). We found that the 25° graphite was about the same stiffness as the Aluminum. We used 25° on the forward keel, 5° on aft keel, 25° on the crossmember and 30° on the 2" leading edges. The tubes were sleeved and bushed wherever there was a bolt or splice with stainless steel bushing or anodized aluminum sleeves. The tubes were joined by sleeves with 2 ton 1/2 hour epoxy glue.

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Very favorable flight tests have been completed on a new Ultralight Products glider. For comparison purposes, UP built their prototype to the same specs as their " 154 " Spyder, but used graphite spars instead of aluminum. The graphite tubing used was made by Graftek/Exxon's Marine Products division. These shiny, dark grey spars are made by wrapping epoxy impregnated graphite material over long mandrels and then firing them under heat and pressure to achieve the proper cure and strength. The result: tubes with half the weight and twice the strength of aluminum! The advantages of this material will enable manufacturers to create smaller gliders having the same lift capabilities as the larger, heavier aluminum gliders. With the reduction in span comes a proportional increase in penetration and quicker control response. The disadvantage of the graphite composite tubes are expense; about \$ 1000 more per glider than aluminum, and a slower fabrication time. Costlier tools are needed; diamond tipped saws and drills must be used to cut and pierce the spars. Construction techniques must be streamlined. According to UP President, Pete Brock, "We are working on a solution to these problems and are confident that we will have the first production graphite Spyders and Fireflys in the hands of the public by next year. " Brock feels that graphite will be to the hang gliding industry what urethane foam was to the surfing industry.

CRAPHITY 15 "050"

240 165

240 165

LOT OF DEPLENION

21HE BOW

WERE BELL

FOR STRONGED

OUT BOMM

11 ULINE DO YIELD TRST 20AD -6061 76 (DRF. 12") WT al Dava Tules tung port DRANUTTIPS 6.5/65 WY OR DAT. BUTTERES 3.51/85 + OVAL 10 165 (INDBATEN AT 30016

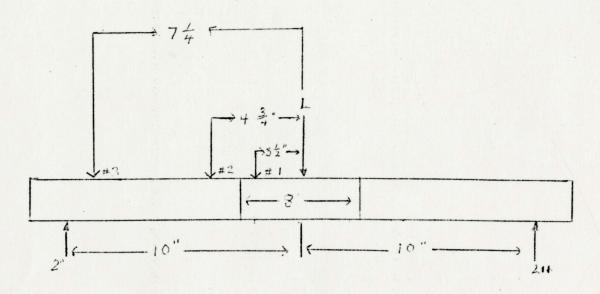
y luminum 150 .003 £ 25° STIPPNECS TOIL WEAVE ANGLE DEX - 1,540 W. T. PET 1.5.70 41/2 602 1.6300. 1.629 1.724 64 " " 1. 800 Jef et un A ?! 1.576 104 2 11 160 1.975 1 AL. 099 7 1.5" DEFI Vo Ve 1.610 A 66 60 1.690-1.767 49 1.834 12 = 1725 1.927 104 MUSER 3 = 9 1.987

6 LBN 1-20p :049h 1004 - \$ 450 DEFLEXION 1.540 STARTING POINT A/165 602 1.05/ 24.165 6 . 547 94 11 6 . 000 1.645" 1.098 216000 4.632 1.695 1.540 4#603. 105 .594 1.6454 64th 60g. Ez 8046.175 100 7038 P=69.37512 R= 1002/01,4 I = 1057602 k PL3 ** MAX = POH 6. PDS

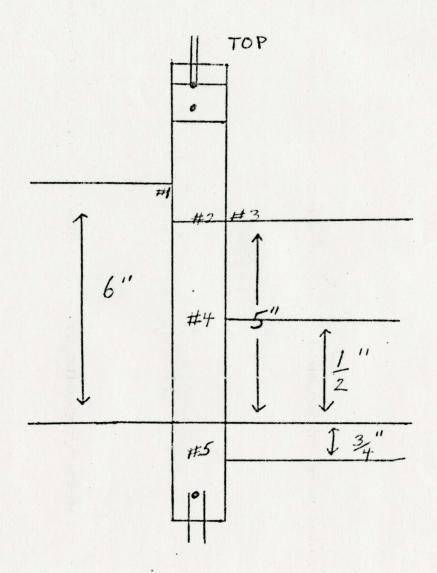
Gage No.	Reading (lbs)	Strain	Micro inches/inch
1	50	35	
1	100	54	
# 1	150	78	
	200	100	
	250	123	1500 678
	50	14	
	100	40	
# 2	150	67	
	200	80	
	250	88	1500 512
	50	30	
	100	52	
[‡] 3	150	78	
	200	95	
	250	118	1500 660
	50	22	
	100	24	
4 4	150	25	
	200	24	
	250	22	1500 306
	50	10	
	100	10	
[‡] 5	150	10	
	200	10	
	250	10	1500 178

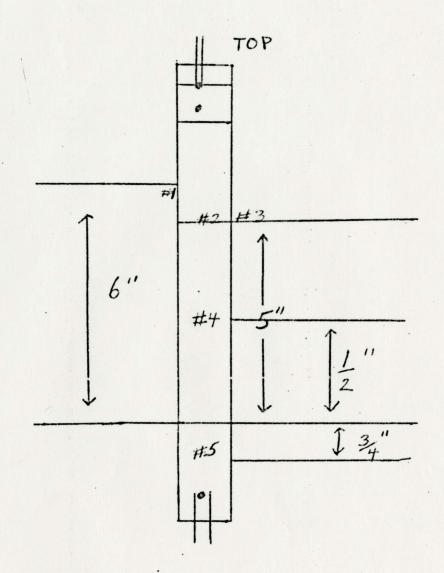
12 x4.080 - . 606 + 300 12 X.006 ± 50 2- CREPHINE NOICES DEF WT 臣= 1.531 1.560 416-602 29 1 17 1.604 94 1.660 E = 78 y Max TIR3 ang T 1.717 69 1.776 89 E - 14.31 × 106 1.810 184 1. 858 E= 14.27 1/0°) 124 4 65 12" .004 + 450 DBA av. 1.59/ 1) 1.626 4/6 60 24 1.862 2-128 94 2.370 64 2.653 74 (E = 2.7×106 2.964 104

Bending Test on 1/2 " OD X .049 at 45°



Gage No.	Reading(lbs)	Strain or comp.	deflection
	50	135	.0155
	100	268	.038
	150	405	.0615
# 1	200	223	.0875
# 1			
	250 300	150 90	.13 .43
	500	94	.76
	50	532	
	100	1080	
	150	1652	
# 2	200	2020	
	250	2549	
	300	3104	
	500	6100	
v ·			
	50	320	
	100	665	
	150	1010	
# 3	200	1360	
	250	1693	
	300	2028	
	500	3430	







University of Lowell
One University Avenue

Lowell, Massachusetts 01854

(617) 454-7811

Test flew glide with new 25° Front head and 5° aft. heels, with 25° Gon tube and 5° aft. heels, with 25° Gon tube had two good long flight glider did feel 4165 fighter.

9. FEB, 80

Test flew glider with new GR leading edger good flight held up good.

ALUMINUM TESTS

1 1/2" X .049"

Weight	Deflection			
0	1.573			
4 lbs 6 oz.	1.610	-	3	
24 lbs 6 oz.	1.690	E=	LP T	
lyl 1bs	1.768	40	y Max i R'avg t	
64 lbs	1.834			
84 lbs	1.927	7 0 0	6	
104 1bs	1.987	E= 8.9	5 X 10	

TUBING COMPARISON

Keel-aluminumWeight--- 1983 grams or 4.37 poundsKeel-graphiteWeight---1180 grams or 2.60 pounds

Cross tube-aluminum Weight--- 1297 grams or 2.86 pounds
Cross tube-graphite Weight--- 877 grams or 1.93 pounds

Leading edge
Aluminum (1 5/8"& 1 1/2") Weight--- 2782 grams or 6.13 pounds
Graphite (2") Weight--- 2603 grams or 5.73 pounds

GRAPHITE TESTS

1 1 2" X .050"+.006 ±5°

reight	Deflection	3
0	1.551	$E=PL_8$
41bs-6 oz.	1.560	48 y Max π R ³ avg. t
-24 lbs-5 oz.	1.604	
44 lbs	1.660	
4 lbs	1.717	•
84 lbs	1.776	E= 14.31 X 10°
104 lbs	1.810	
124 lbs-6 oz.	1.858	

1 1/2" X .050"-.004 ±45°

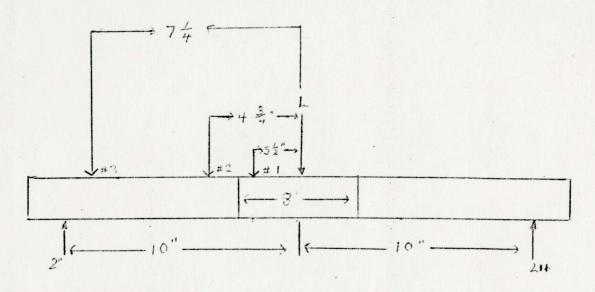
reight	Deflection	
0 4 lbs-6 oz. 24 lbs	1.571 1.626 1.862	E= PI3
44. lbs 64. lbs	2.128 2.370	48 y Maxy Ray t
84 1bs 104 1bs	2.653 2.964	E= 2.7 X 10°

1 1/2" X .050"-.003 ± 25°

Weight	Deflection	
0 4 lbs 6 oz.	1.540 1.570	$E=PL^3$
24 lbs 6 oz.	1.630 1.724	48 y Max n Raavg t
64 lbs 6 oz. 84 lbs 6 oz.	1.800 1.876	E= 8.52 X 10
104 lbs 6 oz.	1.975	

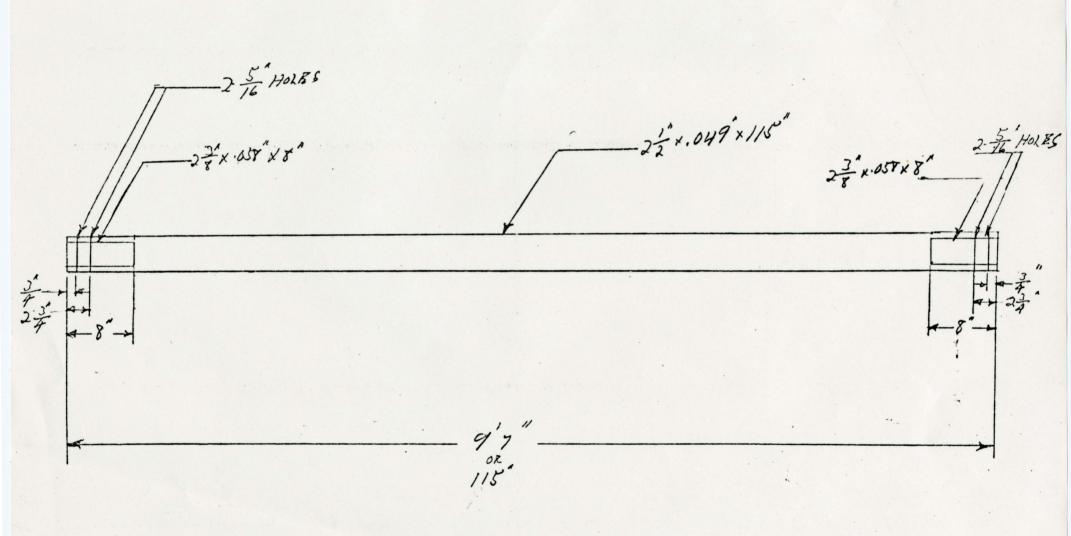
Gage No.	Reading (lbs)	Strain	Micro inches/inch
	50	35	
	100	54	
# 1	150	78	
	200	100	
	250	123	1500 678
	50	14	
	100	40	
# 2	150	67	
	200	80	
	250	88	1500 512
	50	30	
	100	52	
# 3	150	78	
	200	95	
	250	118	1500 660
	50	22	
	100	24	
# 4	150	25	
	200	24	
	250	22	1500 306
	50	10	
	100	10	
# 5	150	10	
	200	10	
	250	10	1500 178

Bending Test on 1/2 " OD X .049 at 45°



Gage No.	Reading(lbs)	Strain or comp.	deflection
		105	0.55
	50	135	.0155
	100	268	.038
	150	405	.0615
# 1	200	223	.0875
	250	150	.13
	300	90	.43
	500	94	.76
	50	532	
	100	1080	
	150	1652	
# 2	200	2020	
	250	2549	
	300	3104	
	500	6100	
	50	320	
	100	665	
	150	1010	
# 3	200	1360	
	250	1693	
	300	2028	
	500	3430	

SENSOR-510-CROSSTUBLE 2- REQUIRED



GRAPHITE TESTS

1 1 2" X .050"+.006 ±5°

reight	Deflection	r- pr ³
41bs-6 oz.	1.550	48 y Max π R avg. t
124 lbs-6 oz.	1.604	
4 lbs	1.717	6
84 lbs 104 lbs	1.776 1.810	E= 14.31 X 10°
124 lbs-6 oz.	1.858	

1 1/2" X .050"-.004 ±45°

Deflection						
1.5)1						
1.626				2		
1.862		E=		PL,		
2.128			48 y	Maxx	R3 avg	t
2.370						
2.653		E=	2.7 X	10		
2.964						
	1.5)1 1.626 1.862 2.128 2.370	1.5)1 1.626 1.862 2.128 2.370 2.653	1.5)1 1.626 1.862 E= 2.128 2.370 2.653 E=	1.5)1 1.626 1.862 E= 2.128	1.571 1.626 1.862 E= PL ³ 2.128	1.5)1 1.626 1.862 E= PL ³ 2.128

1 1/2" X .050"-.003 ± 25°

Weight	Deflection			
0 4 lbs 6 oz.	1.540 1.570	E=	PL ³	
24 lbs 6 oz.	1.630 1.724		48 y Max 7 R2 av	g t
64 lbs 6 oz.	1.800		0 6	
84 lbs 6 oz. 104 lbs 6 oz.	1.876 1.975	E=	8.52 X 10	

TUBING COMPARISON

Keèl-aluminumWeight--- 1983 grams or 4.37 poundsKeel-graphiteWeight---1180 grams or 2.60 pounds

Cross tube-aluminum Weight--- 1297 grams or 2.86 pounds
Cross tube-graphite Weight--- 877 grams or 1.93 pounds

Leading edge

Aluminum (1 5/8"& 1 1/2") Weight--- 2782 grams or 6.13 pounds

Graphite (2") Weight--- 2603 grams or 5.73 pounds

TUBING COMPARISON

Keel-aluminum Weight--- 1983 grams or 4.37 pounds Weight--- 1180 grams or 2.60 pounds

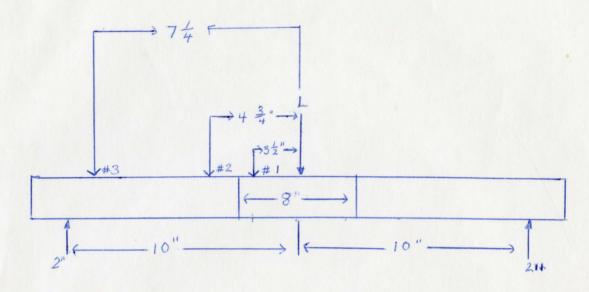
Cross tube-aluminum Weight--- 1297 grams or 2.86 pounds
Cross tube-graphite Weight--- 877 grams or 1.93 pounds

Leading edge

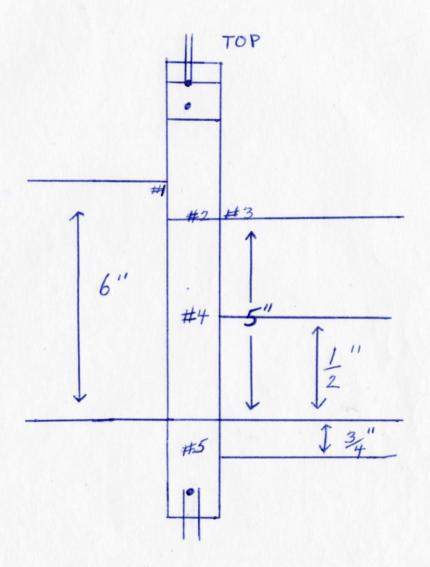
Aluminum (1 5/8"& 1 1/2") Weight--- 2782 grams or 6.13 pounds Graphite (2") & Weight--- 2603 grams or 5.73 pounds

Gage No.	Reading (lbs)	Strain	Micro	inches/inch
	50	35	1500	678
	100	54		
# 1	150	78		
	200	100		
	250	123	1500	678
	50	14		
	100	40		
# 2	150	67		
	200	80		
	250	88	1500	512
	50	30		
	100	52		
# 3	150	78		
	200	95		
	250	118	1500	660
	50	22		
	100	24		
# 4	150	25		
	200	24		
	250	22	1500	306
	50	10		
	100	10		
<i>#</i> 5	150	10		
	200	10		
	250	10	1500	178

Bending Test on 1/2 " OD X .049 at 45°



Gage No.	Reading(lbs)	Strain or comp.	deflection	
	50	135	.0155	
	100	268	.038	
	150	405	.0615	
# 1	200	223	.0875	
	250	150	.13	
	300	90	.43	
	500	94	.76	
	50	532		
	100	1080		
	150	1652		
# 2	200	2020		
	250	2549		
	300	3104		
	500	6100		
	50	320		
	100	665		
	150	1010		
# 3	200	1360		
	250	1693		
	300	2028		
	500	3430		



GRAPHITE TESTS

1 1 2" X .050"+.006 ±5°

Weight 0	Deflection 1.551	E= PL
41bs-6 oz.	1.560	48 y Max π Ravg. t
24 lbs-6 oz.	1.604	
44 lbs	1.660 1.717	
84 lbs	1.776	E= 14.31 X 10
104 lbs	1.810	
124 lbs-6 oz.	1.858	

1 1/2" X .050"-.004 ±45°

Weight	Deflection	
0	1.591	
4 lbs-6 oz.	1.626	3
24 lbs	1.862	E= PL
44 lbs	2.128	48 y Max R R avg t
64 lbs	2.370	6
84 lbs	2.653	E= 2.7 X 10
104 lbs	2.964	

1 1/2" X .050"-.003 ± 25°

Weight	Deflection	
0	1.540	3
4 1bs 6 oz.	1.570	E= PL
24 lbs 6 oz.	1.630 1.724	48 y Max 7 Ravg t
64 lbs 6 oz.	1.800	,
84 lbs 6 oz.	1.876	E= 8.52 X 10
104 lbs 6 oz.	1.975	

ALUMINUM TESTS

1 1/2" X .049"

Weight	Deflection		
0	1.573		
4 lbs 6 oz. 24 lbs 6 oz.	1.610	F=	PT.
The lbs	1.768	4.8	y Max n R avg t
64 lbs	1.834		
84 lbs	1.927	T- 0	95 X 10 ⁶
104 lbs	1.987	E- 0.	75 A 10

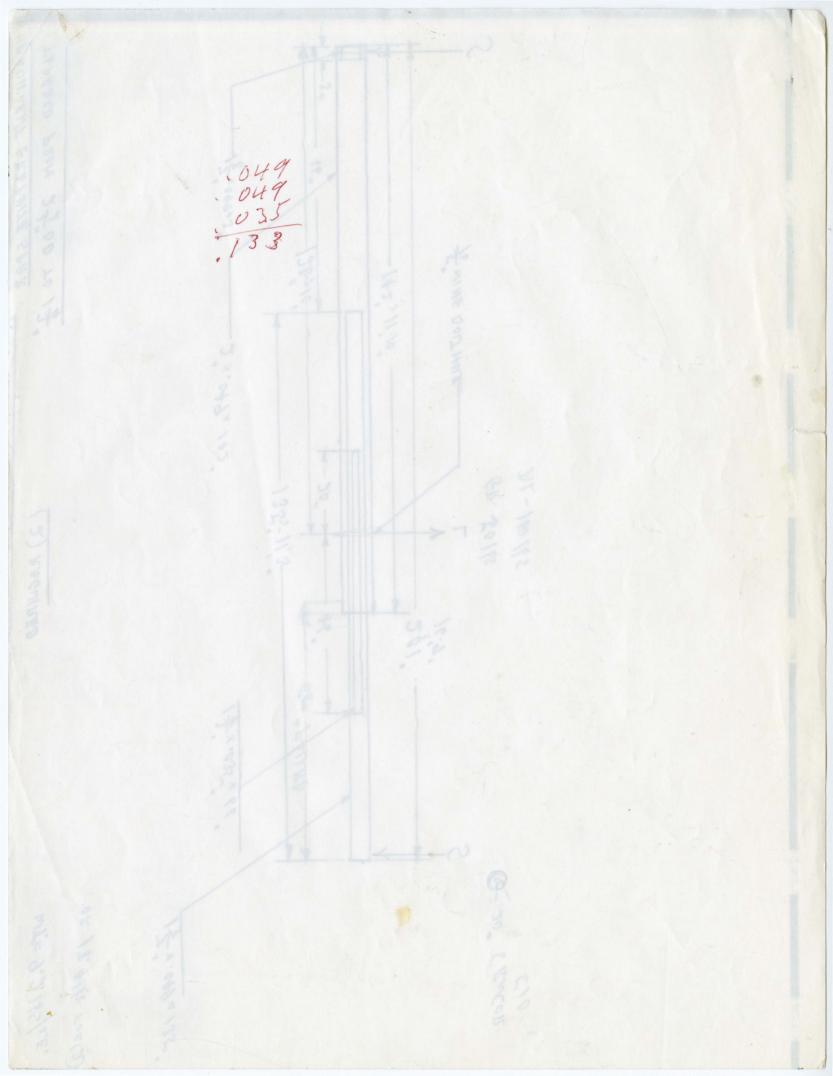
2166 DEOOR3 10-83 3010 50 3 66 76 65 13083 15000 190 0 200 12. 2/000 010 WT OF #2 TUBE # 25.2 5.8 16s WT. OP AL. TURE 9,558 H.

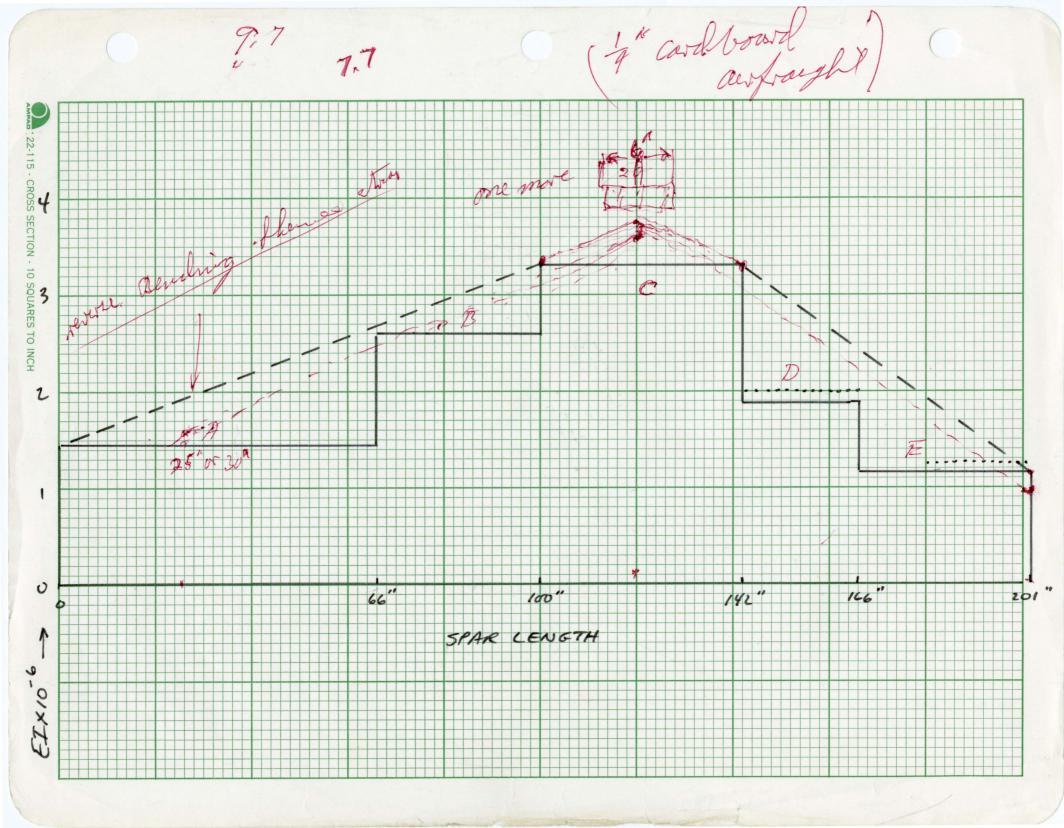
FORD = 25 SENSON-510 CROSS- TUBE: "2- REQUIRED EAL = 10 x /06 PS/ 106 = 34 (TT (D+-d 4) (D4-d4)= = 10 (D4-d4) AL. D4-24-X d = D4- H 100 - 14) = x (25) - (2.402) + =x 39.06 - 33.62 14 5.39 71-11.585-11 24=D4-K K= 1.505 24 = (2.5) 4 - 1.584 14 = 39.06 - 1.584 = 37.476 d = (37.476)4 d = 2.474 x1

8D.GR = 2.500" OD 1.0. 2.474 40 THOWALL = .013"/

Glider Siz	Alider Size Comet 135						Date 4-10-81		
X-Bar Material / 7/8 x . 058						innersleeve 13/4 x-049			
	. A	В	_		, in the second			Ç	
		+		-			A,	+	9
	-3	3"-		,	INCESI CE NNBR SL			-3	
			Hole		90/2,- Size		Distance		
	A	Plate	At Port	1	3/8	TOP	3/4	ef tuse	
	В	PLATE	At Root		3/8	Top	2 1/8	11	
	C,	F.H.	ak		1/4	Top	3/4	Fr GHO of tose	

BRUIVERNT GRAPHITE SPAR (2) REQUIRED WT= 9.2165/2E. TAPEARD FROM 2500 TO 13" OR 18.4165 FOR(2) 17 x.049 x 135 13×1035×66 -2"x.049x 142" -18 × 099+3" 1355-11,3-120:10 4-20" -142', 11:10" 201" 5 WING BOLTHOLE GR. 50165 @ 20° SRNCOR AL-100/65 510

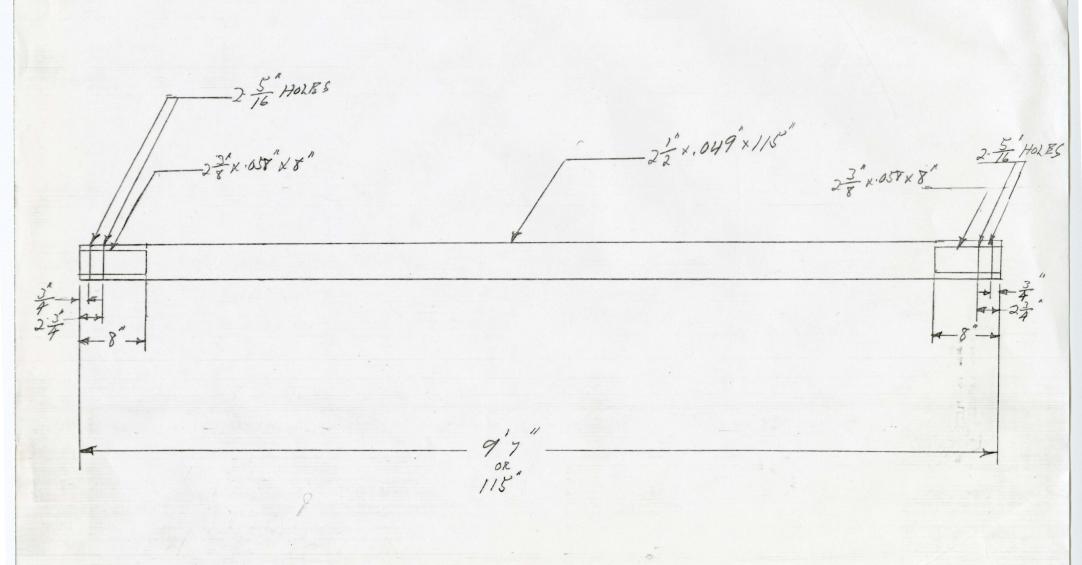




SENSOR-510-CROSSTUBLE

2- REQUIRED

AL-6061-76



GRAPHITE 2 x.050 WALL AL-12 00. 058 WALL DRF. DBF WT. 0 .195" 10/68 10/65 . 430 .905 4 20 " 20 .610" .629 4 30 4 -30 .782" - .767 " 90 m 40 .959" .910 " 50 -20 = 1.065" 1.132 60 " 60 1.230 - 1.297 70 " 70 1.467 1.388 88 n 88 1.546 1.633 90 " 96 1.710 -1.820 100 2 180

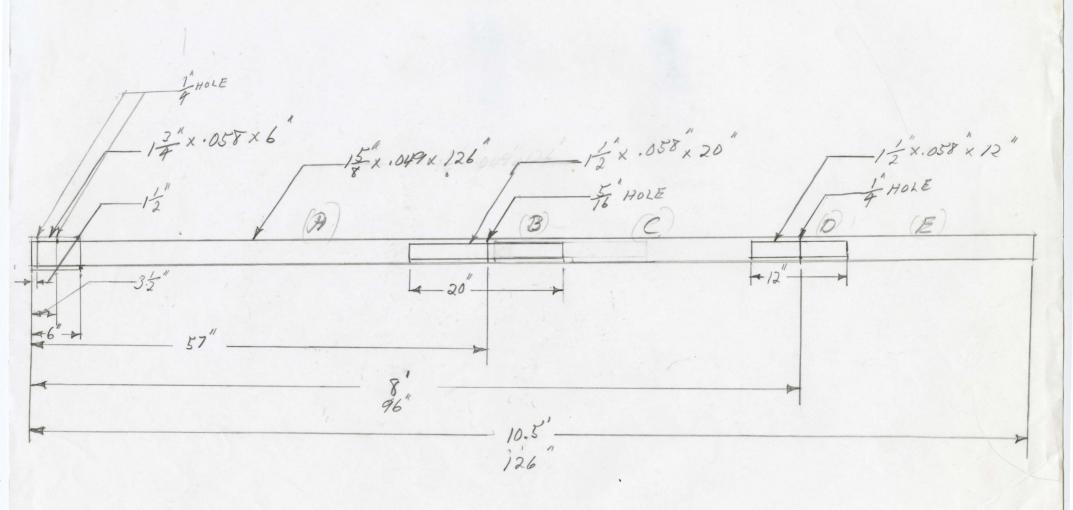
DIF. IN DEF .. 110"

ACT. WT-W/ BORTA TIP DOP 9. 7165 WT.= 9.2 165/2E. EQUIVERNT GRAPHITE SPAR (2) REQUIRED TAPEARD FROM 2500 TO 13" OR. 19.4 165 FOR 2 002.1" = 3.75 165 = 2 = 7.50 165 - 19.4 - 7.5 = TWT.S. \$11.9716 OR 18.4165 FOR(2) TOTAL EST WY. SAVINGS/ 20160) 00,1.725"=5.75 165. 15 16 s 17 x.049 x 135 75-80% GRAPHITE 20-25/5=GLASS .2"x.049x 142" 13 x .035 x 66" 18 1049+3" 46 FAILAD. NERD_SAME +HOOP STREWTH FRP 39-409 4.4 GR -> 50 765 -> FAIL 3.6 5 WING BOLTHOLE KING POST & DOWN TREL DL WTAPX = 4.37 160 AL - 100 16 E=34 X10 psi TUBRES TOTAL WS = (12.62 /8) AL WT = 1.5160×4 IWID= 6.0/85 En= 10×10 psi 16.W = 2.4/65 CROSS TUBE WS = (3.6)65 BATTER WT AL = 5.8 160 9½ × 2½ × .049 (1388 165-43") WTG. = (2.32 160) 2=CT. WTA = 24x9 = 2x3.58 = 7.16165 W.S. 73.48 /65 WT.6 = 3.86 HS (WS = 3.3 165) WT SRNSOR 510 16C = 66 Ha

SENSOR 5/0 KEEL

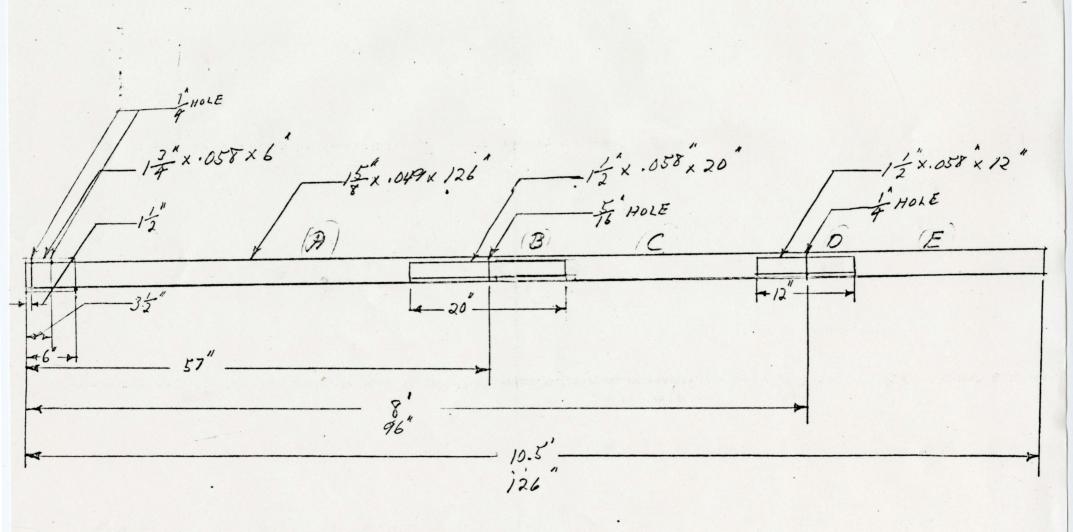
1-REQUIRED

D2-6061-TEUTPAN = .285



WT = 3.13 160 13 Din

SENSOR SID KEEL



TAPEARD FROM 2500 TO 13

SUPLIE 5 = 1'
WT = 9.2165/2E,
OR 18.4165 FOR(2)

